

# Morphing in aerospace: aims, challenges and current showstoppers

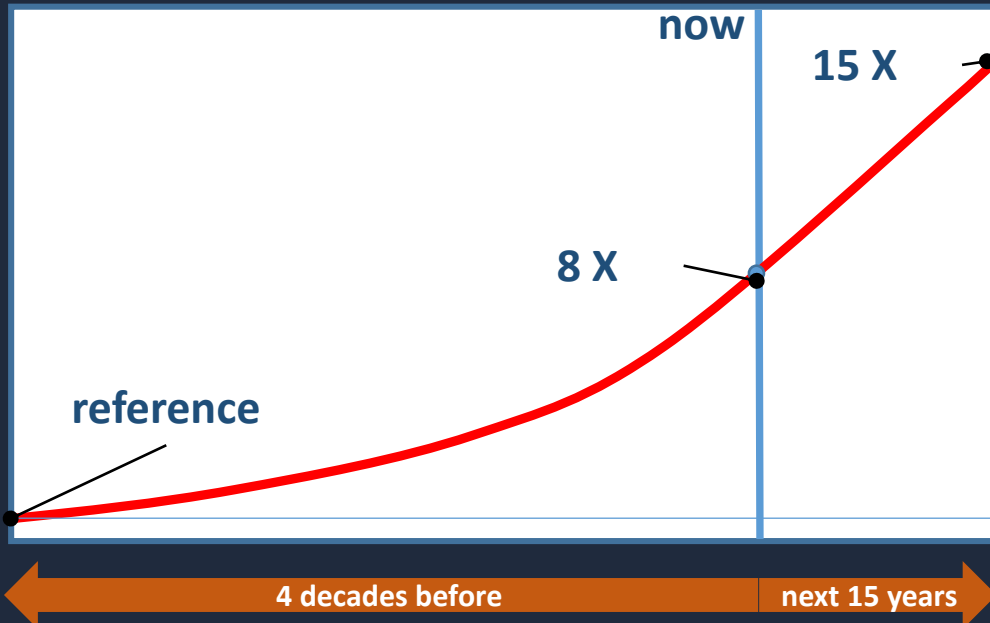
*S. Ameduri, A. Concilio, the Italian Aerospace Research Centre*



**Let's start with... MOTIVATION**

### MARKET

Revenue passenger kilometers

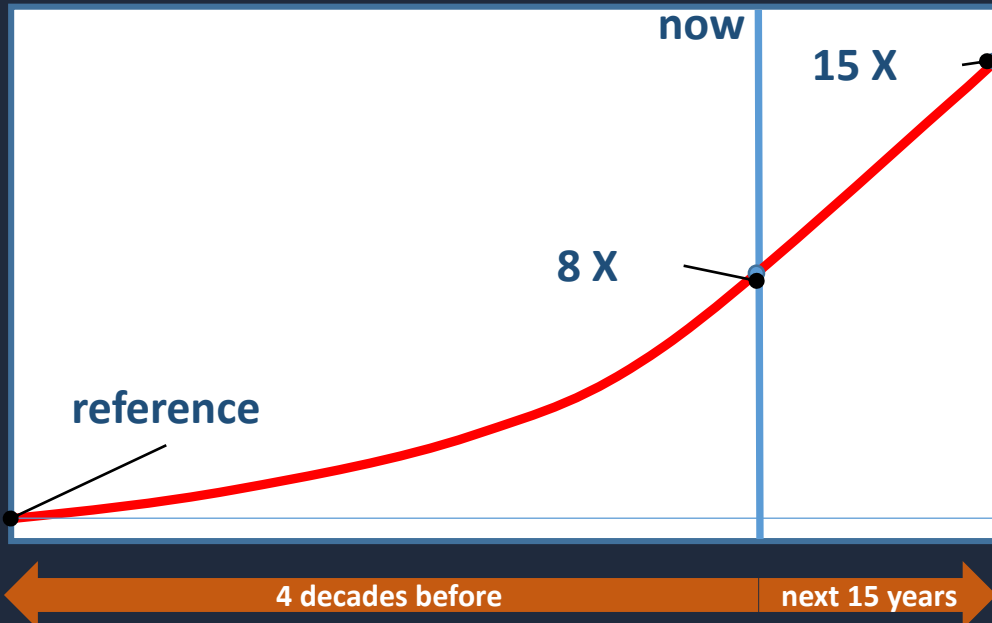


<https://www.raconteur.net/business-innovation/aviation-industry-soaring-into-the-future>

- air navigation agencies employ 7.6 million people
- aviation supports 58 million jobs and contributes \$2.4 trillion to the world economy

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Revenue passenger kilometers



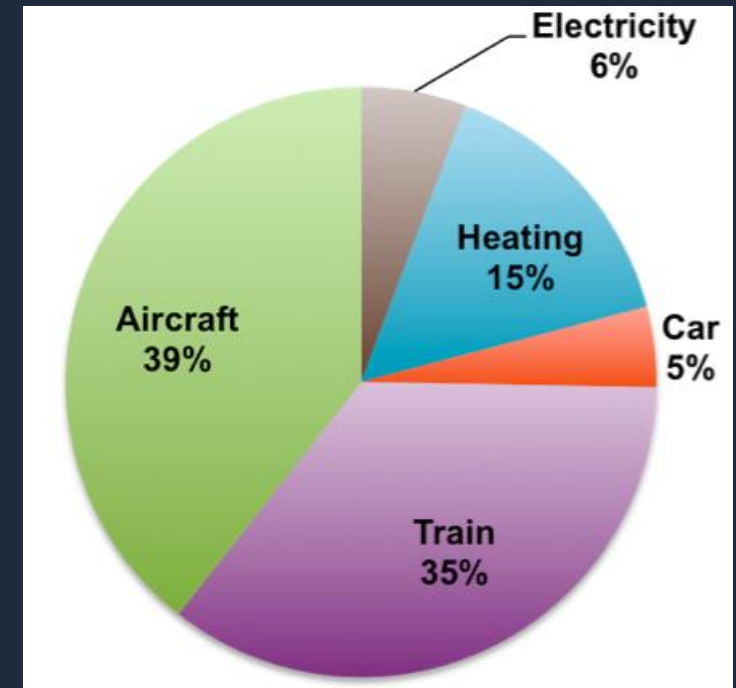
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## ENVIRONMENT

Environmental impact goals for the 2050 (ACARE agenda):

- Cutting CO2 emission by 75%
- Reducing NOx by 90%
- Reducing perceived noise levels by 65%



<https://theicct.org/blogs/staff/carbon-diary-reluctant-traveler>



**AERODYNAMICS**

**PROPULSION**

**ON-BOARD  
SYSTEMS**

**STRUCTURES AND  
MATERIALS**



among the other technologies... morphing...

among the other technologies... morphing... why?

*“a multidisciplinary technology that combines integrated aerodynamic, structures and on-board systems knowledge”*

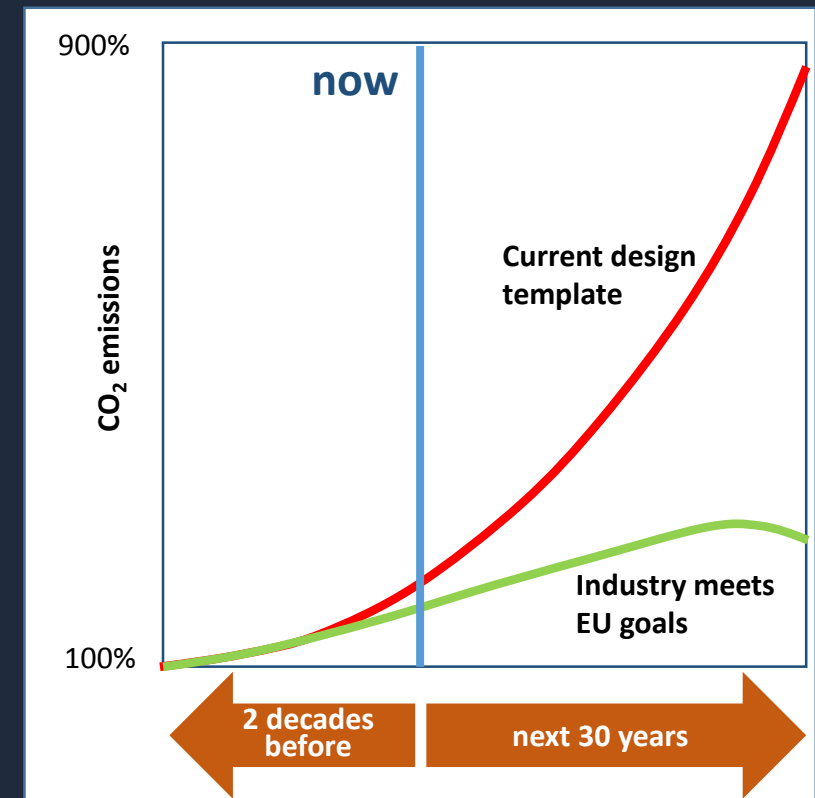


## among the other technologies... morphing... why?

- Higher L/D → fuel saving → lower costs and emissions
- High lift generation → take-off/landing performance
- LC&A → reduced stress level → more safety, lighter main structure, lower running costs
- Lighter, less complex and more reliable architectures

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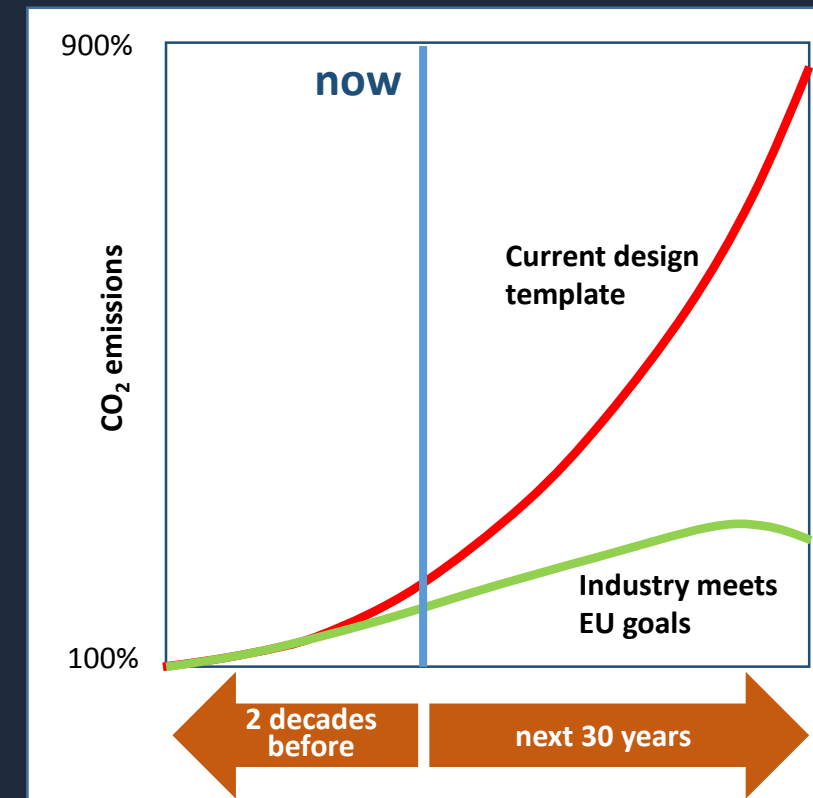
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**Efficiency increase higher than 20%, in synergy with other technologies**

Unfortunately... not all that glitters is gold!



Herein some relevant problems are shown:

**Load transmission to the supporting structure**

**Scalability**

**Strength vs flexibility skins**

**Dynamic/aeroelastic response**

**Load transmission to the supporting structure**

**Scalability**

**Strength vs flexibility skins**

**Dynamic/aeroelastic response**

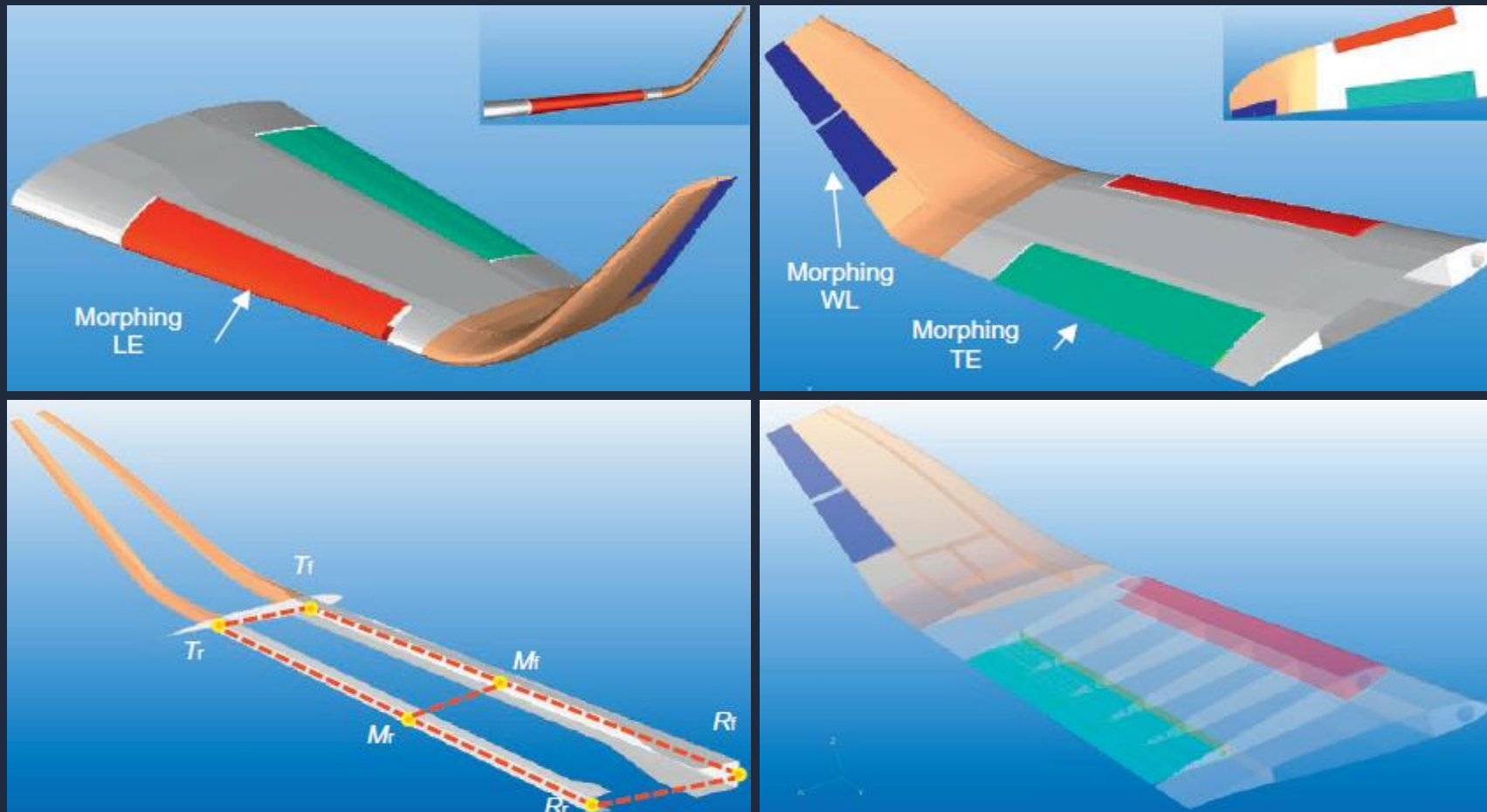
Morphing systems may be classified as **kinematics and compliant**, each with its own characteristics and peculiarities. For instance, kinematic systems are usually related to a larger number of parts and components (architectural complexity), while compliant systems generates larger localized stress and strain fields (design, materials).

In spite the different issues herein addressed should be specialised for each of the abovementioned classes, the main features shall have a general validity unless explicitly stated.

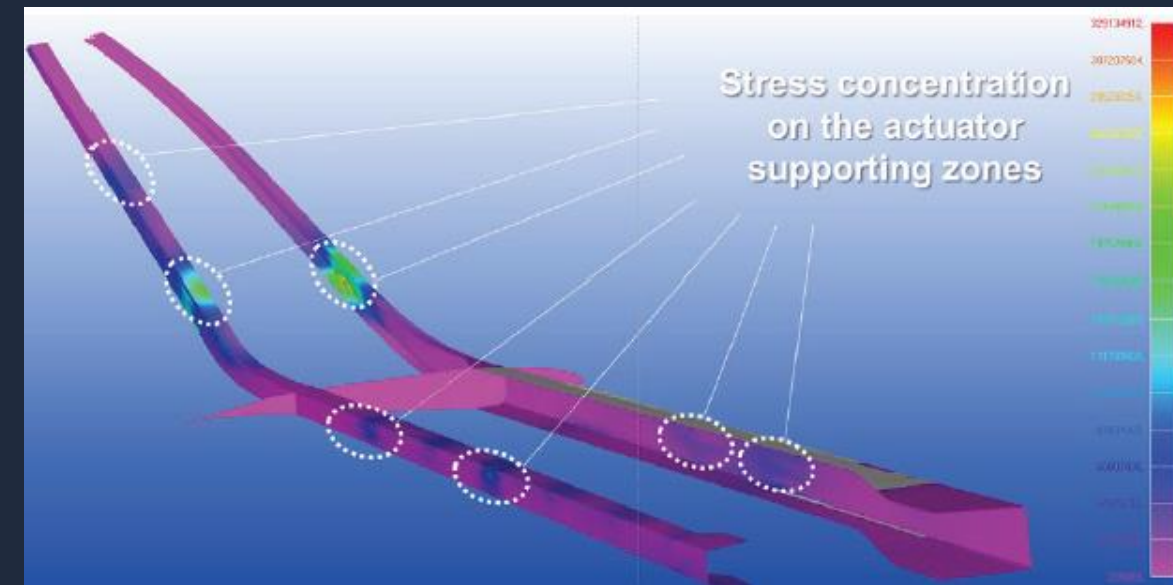
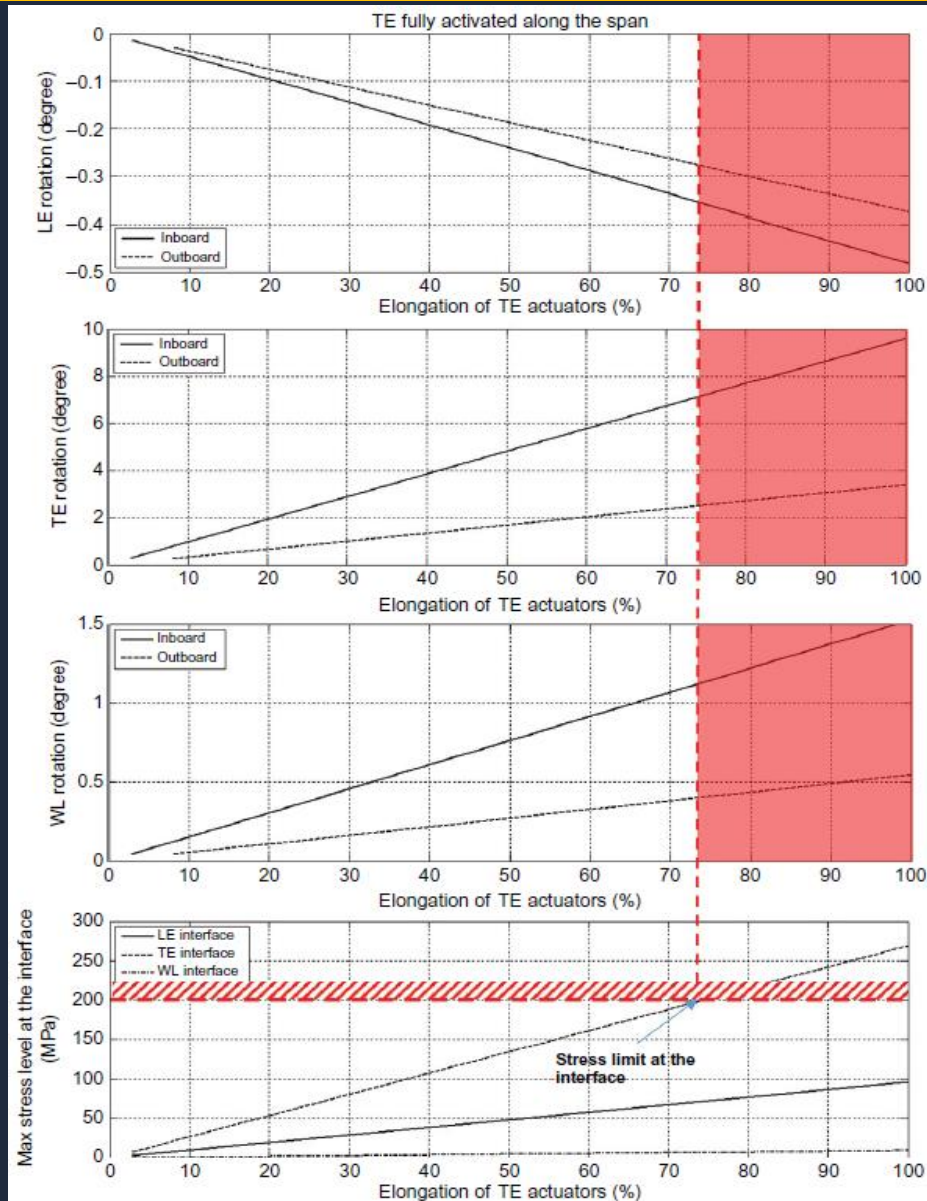


# Load transmission to the supporting structure

## Wing layout, SARISTU



### Effect of the elasticity of the interface and cross related deflections

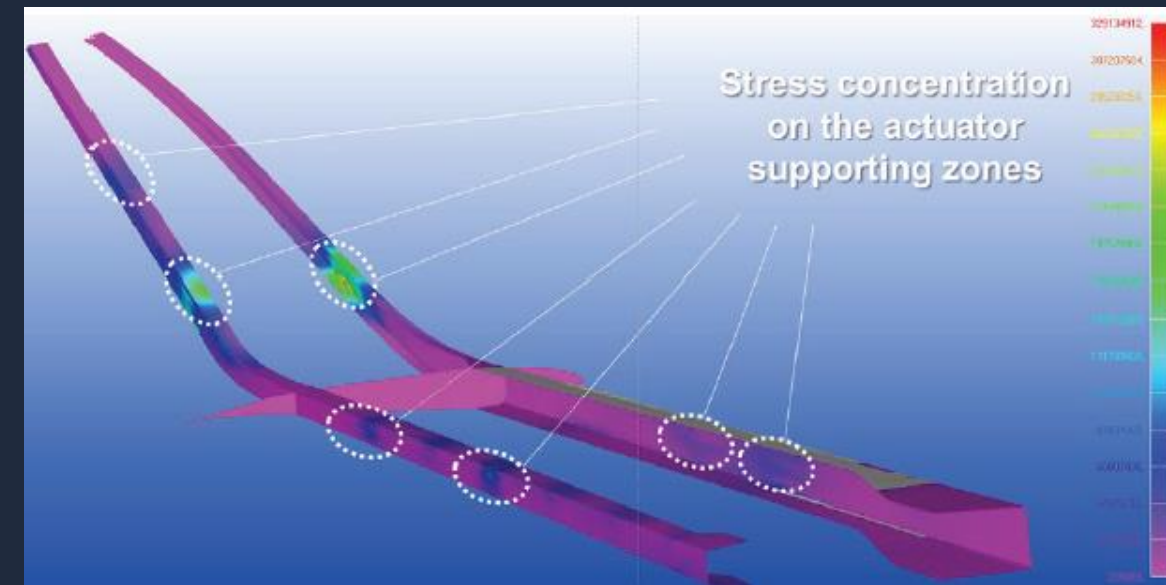






Relative Ratio between Induced and Actuation Rotations %, for each Morphing Device

	Actuated		
	LE	TE	WL
LE	1.00E+00	7.05E-02	3.33E-08
TE	5.52E-04	1.00E+00	5.48E-08
WL	2.19E-03	1.76E-01	1.00E+00
Wing	3.93E-03	4.53E-01	1.38E-04

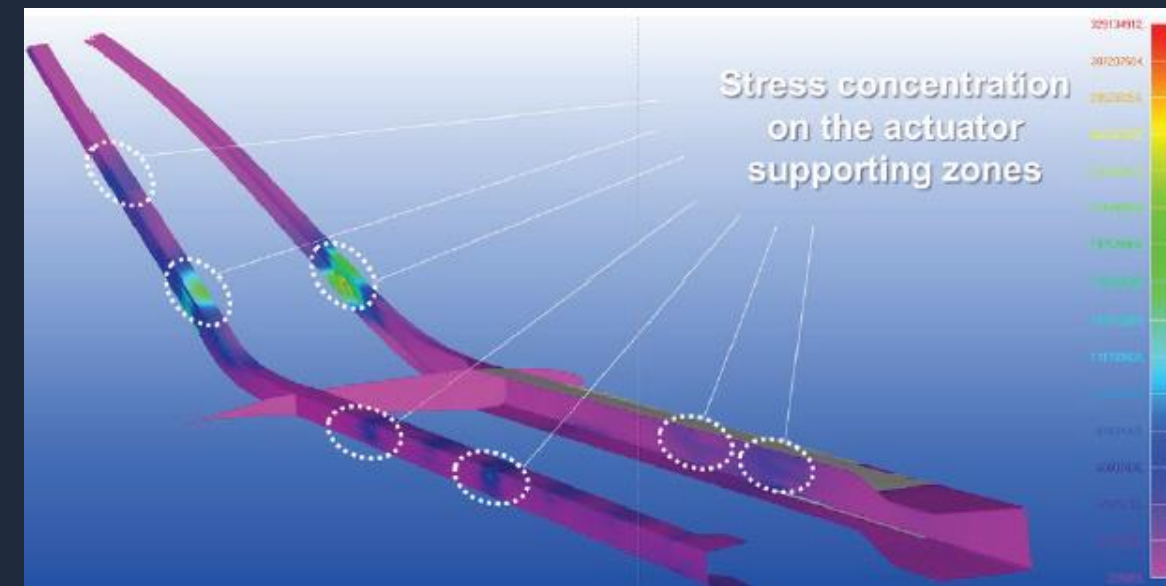


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## Load transmission

- Adaptive structures transmit significant loads at the interfaces
- Part of those loads comes from the elastic reaction of the structure when deformed
- This aspect has impact in terms of:
  - stress accumulation at the interfaces
  - losses of functionality of the morphing device
  - alteration of the functionality of other attachments

## Strength vs flexibility

- Different parts of a morphing structure undergo large deformations and, at the same time, must withstand high loads
- Among the other components the skin is particularly impacted... in fact:
  - it is the most centrifuged component with respect the others with a consequent amplification of displacements and deformations
  - it must keep its shape under the aerodynamic loads

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thus let's just focus on the  
**SKIN**

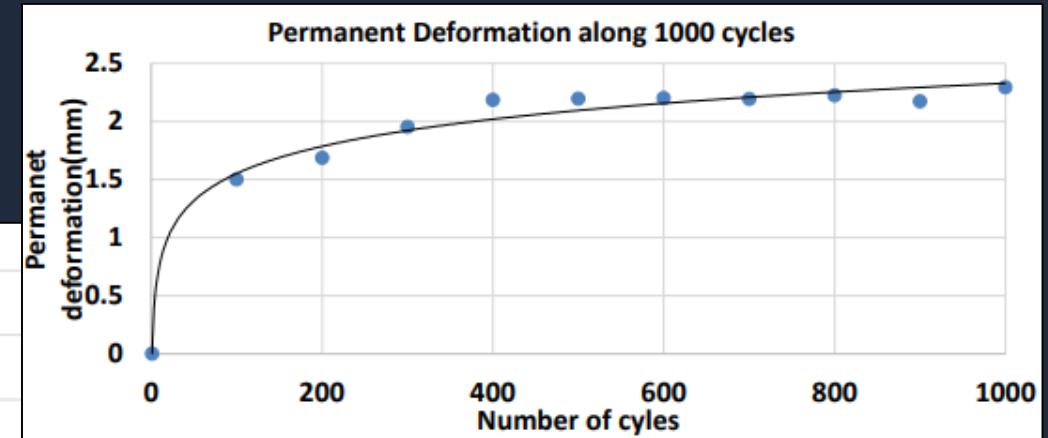
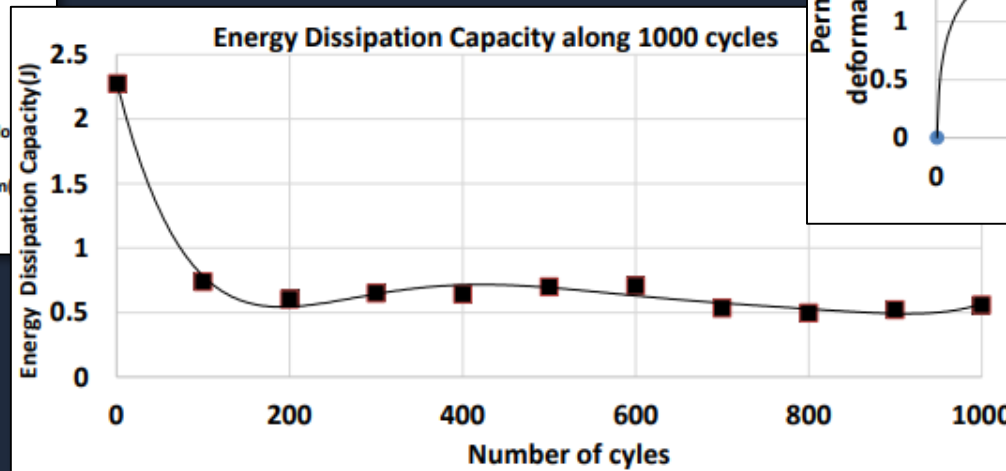
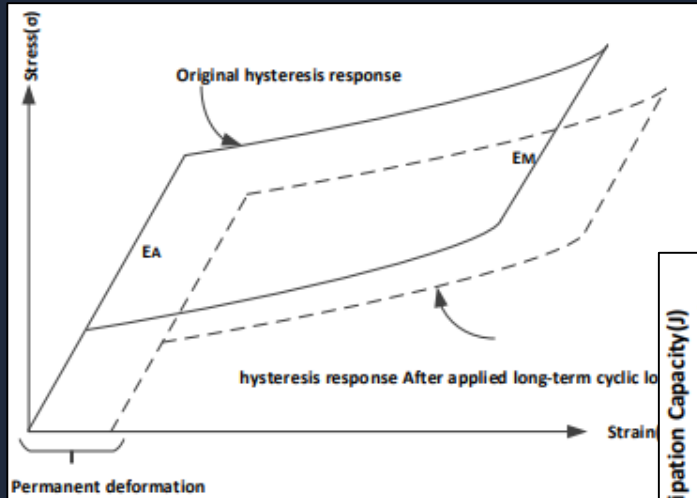
## Skin requirements

- elastic/flexible in chordwise direction to allow low force actuation
  - stiff in spanwise direction to withstand aerodynamic and inertial loads
  - toughness
  - abrasion and chemical resistant
  - resistant to weather agents
  - high strain capability → superleasticity
  - Aging and fatigue resistant
- 
- *Kikuta, M.T. Mechanical properties of candidate materials for morphing wings, 2003, p 123, Department of Mechanical Engineering, Virginia Polytechnic Institute and State University*
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# Candidate materials and architectures

## Superelastic materials... SMA

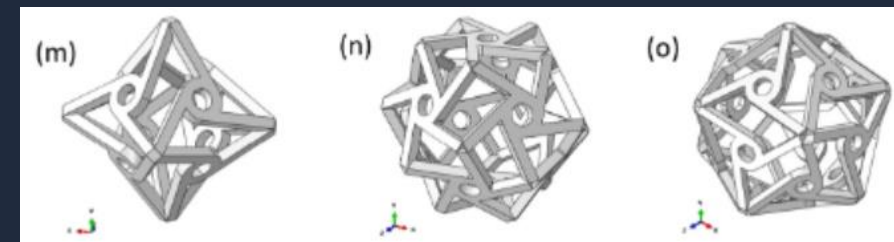


*S Zareie, MS Alam, RJ Seethaler, A Zabihollah, "Effect of cyclic loads on shape memory alloy-based component of cable-stayed bridge", CSCE Annual Conference, Laval (Greater Montreal), June 12 - 15, 2019*

# Candidate materials and architectures

## Auxetic materials

- advanced manufacturing techniques are needed for fabrication of composites based on chiral metamaterials
- research attainments on design optimization does appear still limited, especially for 3D chiral metamaterials
- mechanical properties of 2D and 3D chiral mechanical metamaterials under larger deformations should be further investigated
- more theoretical models should be proposed for catching the multifunctional performances of 2D and 3D chiral metamaterials, such as: impact energy absorption
- more experimental verifications should be performed for assessing potential industrial applications

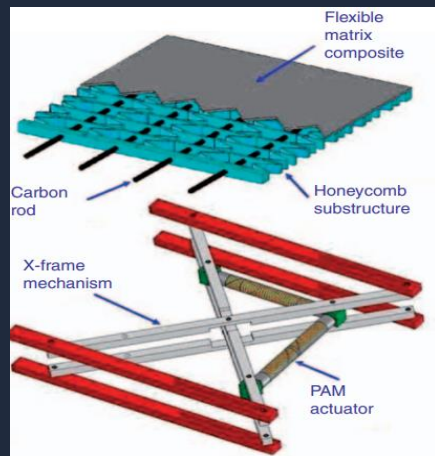




# Candidate materials and architectures

## EMC

- Supporting structure required
- Low out-of-plane stiffness



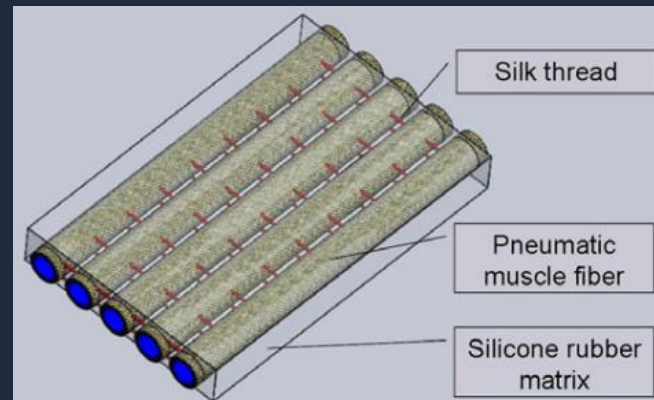
## Laminated composites

### Honeycombs between stretchable laminae

- Composite thickness

### Compliant laminae with sandwiched corrugations

- Composite thickness



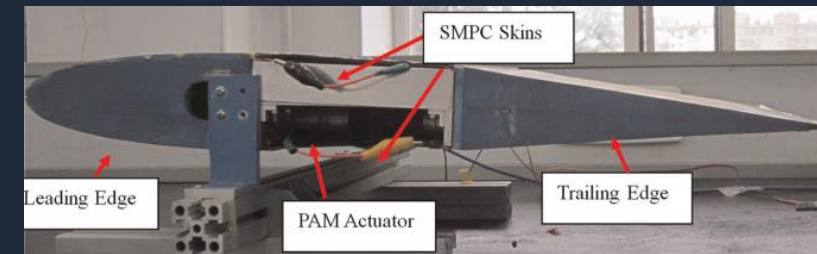
*Yijin Chen, Weilong Yin, Yanju Liu, Leng Jinsong, "Structural design and analysis of morphing skin embedded with pneumatic muscle fibers", Smart Materials and Structures 20(8):085033*

### Fiber reinforced with SMPs

- Low actuation performance

### Fluid channels between compliant media

- Bulky fluid sources



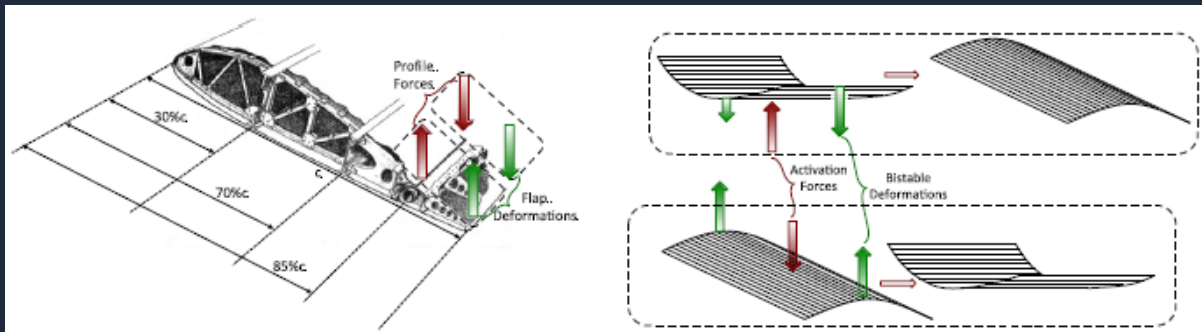
*Jian Sun, Yanju Liu, Jinsong Leng, "Mechanical properties of shape memory polymer composites enhanced by elastic fibers and their application in variable stiffness morphing skins", Journal of Intelligent Material Systems and Structures, 2015, Vol. 26(15) 2020–2027*

*E.A. Bubert, B.K.S. Woods, K. Lee, C.S. Kothera, N.M. Wereley, "Design and Fabrication of a Passive 1D Morphing Aircraft Skin", J. of Intelligent Material Systems and Structures, Vol. 21—November 2010*

*V. S. C. Chillara, M. J. Dapino, "Review of Morphing Laminated Composites" Applied Mechanics Reviews, JANUARY 2020, Vol. 72 / 010801-1*

# Candidate materials and architectures

## Multi-stable composites



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- low load bearing capability of some stable states
- temperature-dependence of material properties  
NON-UNIFORM temperature environment
- accurate mathematics modeling of complex nonlinear small- and large-amplitude dynamics
- control of snap-through and snapback motions
- optimizing stiffness in load and actuation directions to ensure structure compliance

## Scalability



- For the sake of clearness we will assume a geometric scale factor of 1/1000, thus passing from m to mm
- If we want to have the same stiffness for the original and scaled structures the relation  $K = \frac{F}{x}$  must hold for both of them
- However, for the present case,  $K = \frac{EA}{L}$ ... this means that keeping constant the material, there is a factor of 1000 between the original and scaled structure

## Scalability



- Finally, dynamic behaviour:  $\omega = \sqrt{\frac{E}{\rho L^2}}$
- Referring to the same material, the ratio  $E/\rho$  does not change, while the length is reduced by 3 magnitudes
- Consequently, the frequency of the scaled structure is 1000 times the original one

# Scalability

Another example... compliant mechanism vs size/weight/load

Parameter	Herti / AR4 (ref)
Size (chord)	~8
Wing loading	3
MTOW	160

Considering a metallic honeycomb filler for the wing and the limitations given by the wing depth and chord, a weight ratio of **500** results between the two wings

## Scalability

- Morphing structures are very scale-dependent
- Different architectures must be adopted to actuate the same functionality on geometrically similar models
- For WT tests, it is not possible to define a unique scale factor for geometry elasticity and aerodynamics (...and aeroelasticity...)
- Compliant mechanisms appear non-convenient as the aircraft size arises



## Aeroelastic behavior of morphing systems

- Morphing systems enable more DOFS with benefits in terms of LCA efficiency
- However
  - Due to the intrinsic non linearity (large displacements and related stress accumulation) dedicated modeling approaches are needed, encompassing clean and deformed configurations
  - More distributed dofs → higher modal density → higher probability of bending and torsional modes coupling and higher probability of failures

*Claudia Y. Herrera, Shun-fat Lung, Gregory Ervin, Peter Flick, “Aeroelastic Airworthiness Assessment of The Adaptive Compliant Trailing Edge Flaps”, SFTE 2015 Symposium*

*Maria Chiara Noviello, Ignazio Dimino, Antonio Concilio, Francesco Amoroso, Rosario Pecora, “Aeroelastic Assessments and Functional Hazard Analysis of a Regional Aircraft Equipped with Morphing Winglets”, Aerospace 2019, 6, 104*

## Conclusions

An overview of the challenges and current showstoppers of the morphing technology was provided; we found that:

- Morphing systems exhibit an unconventional transmission modality of loads, due to the elastic reaction of the deformed structure
- The strength-flexibility paradox primed the development of different types of skins, characterized by specific limits
- Morphing systems cannot be easily scaled → no immediate transportation of the technology to different aircraft classes and dedicated/partial requirements for WT tests
- Morphing systems exhibit specific aeroelastic behavior, due to their intrinsic non-linearity and the higher dofs with respect to conventional systems



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**Thank you  
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